

Local Buckling Rules for Rotation Capacity

M. L. DAALI and R. M. KOROL

ABSTRACT

Structural frames designed or proportioned to resist seismic forces must possess adequate ductility to redistribute internal forces or have needed energy absorbing capability. This paper addresses the interaction effects of a steel member's plate slenderness values and its lateral slenderness on rotation capacity. It is shown that members with slenderness values close to the limits specified by the codes of practice may not be able to redistribute moments adequately under seismic loading. A critical value for the lateral slenderness is proposed to be used with current local buckling rules prescribed in North American standards for a structure to develop its full ductility potential.

INTRODUCTION

A desirable attribute of a steel beam subjected to static loading is the maintenance of its full moment capacity under extensive plastic deformation. Such response is perhaps best exhibited in the member's moment-curvature relationship. In recent design codes, members classified as plastic design sections are assumed to sustain a resisting moment without loss of strength until a required rotation capacity is achieved, thus permitting full redistribution of bending moments to other parts of the structure. Figure 1 is indicative of this behaviour. The response of such members to transverse loading is generally well understood up to the initiation of local buckling. However, concern arises about its ability to achieve adequate plastic rotation in the inelastic range. A number of researchers have pointed out weaknesses of some structural elements that satisfy the requirements of plastic design. Among the points mentioned are the rotation capacity and premature local buckling of the members. Thurston¹ expressed concern about the small safety margin implicit in the New Zealand code limits for flange slenderness ($b/2t$) and web slenderness (h/w). His opinion was based on earlier than anticipated local flange buckling in the testing of classified plastic design beam sections that resulted in premature moment capacity deterioration. As such, he proposed that the ($b/2t$) ratio limits be reduced. Beamish² noted meanwhile, that under cyclic loading, local flange buckling can occur at

relatively low ductility in members having plate slenderness ratios close to the plastic design limits. Mitani et al.³ and Takanashi et al.⁴ highlighted the poor ductility achievement of some members complying with the requirements of plastic design. Kemp⁵ pointed out the possibility of some steel sections with less than adequate ductility for plastic analysis; he mentioned in particular the influence of the interaction between the different modes of buckling, not considered in existing codes. More recently, Ghobarah et al.⁶ pointed out that the slenderness limits ($b/2t$) and (h/w) for beams in MRFs in highly seismic areas may be, in some situations, unconservative thus leading to a reduction in load carrying capacity.

The flanges and webs of a beam tend to behave as plate elements.⁷ It is well known from the post-buckling theory of plates that, even when distorted after local buckling, plates having edge restraint are still able to sustain significant levels of load. This has been observed through beam bending experiments showing that considerable rotation capacity exists after local buckling is observed.⁸

Frequently, local flange buckling is observed simultaneously with lateral buckling. Alternatively, it has been noted that an interaction of local flange buckling with local web buckling may occur also. In this context, Kato^{9,10} proposed interaction relationships relating flange and web slenderness; however, no lateral slenderness was included in the analysis. About the same time, Kuhlmann¹¹ showed that the rotation capacity is governed by three parameters—flange slender-

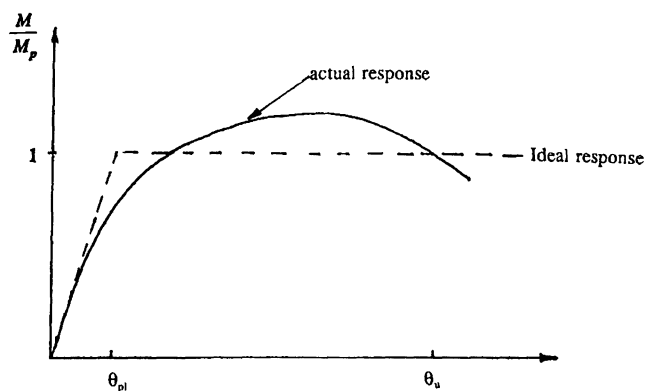


Fig. 1. Behavior of Class 1 (plastic design) sections under monotonic loading.

M. L. Daali and R. M. Korol are with the Department of Civil Engineering Mechanics at McMaster University, Hamilton, Ontario, Canada.

ness, web slenderness, and the steepness of the moment gradient, defined in terms of unsupported length over the width of the section, $L/2b$. The interaction of the different phenomena mentioned above is of a highly complex nature. As far as design standards are concerned, the Japanese code¹² has attempted to address this issue, while other codes do not cover combined responses even for monotonic loading. For example, slenderness limits for flanges and webs do not reflect combined buckling phenomena. Thus, making design rules simpler may result in potentially inaccurate member selection.

With the aforementioned in mind, a procedure accounting simultaneously for flange and web plate width or depth to thickness ratios and lateral slenderness is proposed. Critical values of normalized flange and web slenderness ratios are obtained in accordance with the normalized lateral slenderness and the required rotation capacity for a condition of plastic design. A tool is offered to researchers and design engineers for defining maximum plate slenderness ratios (flanges and webs), and lateral slenderness requirements to achieve a given amount of rotation capacity.

REQUIRED ROTATION CAPACITY

For a plastically designed member subjected to bending moment and laterally braced in accordance with the specification in use, the available rotation capacity, R_u is a ductility parameter that determines how effectively internal moments can be redistributed once M_p is reached at a given cross section. It is here defined by the amount of total rotation beyond the plastic limit that can be sustained before unloading below M_p takes place and is herein given by:

$$R_u = \frac{\theta_u}{\theta_{pl}} - 1$$

In general, steel members of an open style cross section may fail by either local buckling or lateral buckling or both. Because of requirements in design to prevent lateral buckling, the most important aspect which governs the rotation capacity of a member is its potential for local buckling. Since the latter mode is crucial, the width-thickness ratios of compression elements will dramatically affect the delivered rotation capacity.

Recently, Bild and Kulak¹³ compared rules governing local buckling of plastic sections for 13 different specifications. They noted that the modified flange slenderness, $(b/2t)\sqrt{\sigma_y}$, expressed in SI units, ranges from 131 to 174. Meanwhile, the modified web slenderness, $(h/w)\sqrt{\sigma_y}$, was observed to range from 981 to 1,365.

In seismic design, there is no agreed-upon value for the required rotation capacity. The situation is complicated further by the fact that there are not only different types of ductility to be considered, but also the notion of force modification needs to be embraced. Popov¹⁴ distinguishes among the material capacity, R_m , the member capacity, R_u , and the

system capacity, R_s , and suggests the relationship that $R_m > R_u > R_s$. Hall¹⁵ recommends employing an R_u value of $1.5R_s + 0.5$. Meanwhile Chopra and Newmark¹⁶ suggest that it is prudent to provide a required rotation capacity R_u of six to eight while Fukumoto and Itoh,¹⁷ quoting from AIJ LSD¹² propose a value of four. Meanwhile, Coşenza et al.¹⁸ link a member's rotation capacity, R_u to the system ductility, μ_s . They suggest a value for R_u of $(\mu_s - 1) / \psi$ with ψ fluctuating between 0.36 and 1. The current Canadian Code¹⁹ for the design for fully ductile moment resisting frames employs a force reduction factor of four for a system ductility μ_s . If Cosenza's suggestion is to be used, and with $\psi = 0.5$ representing an average of the maximum and minimum suggested values, a required member ductility of six, corresponding to the minimum value suggested by Chopra and Newmark, would then apply for seismic design in Canada. A graphical representation of the system ductility-rotation capacity relationship is given in Figure 2.

In the United States, the force reduction factors R and R_w for NEHRP²⁰ and UBC²¹ respectively are formulated²² by the expressions $R = R_\mu \Omega$ and $R_w = R_\mu \Omega Y$, where R_μ is the ductility reduction factor, Ω is the overstrength factor, and Y is a factor evaluated at 1.4. Newmark and Hall²³ relate the ductility reduction factor to the system ductility as follows:

$$R_\mu = \mu_s \text{ for velocity and displacement amplification regions.}$$

$$R_\mu = \sqrt{(2\mu_s - 1)} \text{ for acceleration amplification region.}$$

The standards NEHRP and UBC do not explicitly specify the value of the overstrength factor. This leaves the possibility of combining the system ductility, μ_s , with the overstrength factor in various ways. If, however, the overstrength factor is chosen to have a value of 2.3 as suggested by Uang, then the force reduction factor of NEHRP and UBC, gives for fully ductile moment resisting frames, average system ductility values, μ_s , of 3.6 in the velocity region and 6.9 in the accel-

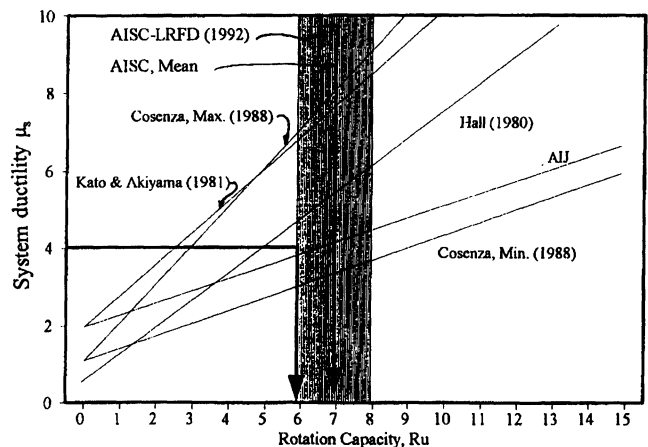


Fig. 2. System ductility—rotation capacity relationship.

eration region. If Cosenza's suggestion is to be used, an average required member rotation capacity, R_u of about 4.9 and 11.1 would result for the velocity and acceleration regions respectively. These R_u values represent clearly a much wider range than range six to eight suggested by AISC. For member assessment, we will, however, simply use an average AISC value of seven in the United States.

AVAILABLE ROTATION CAPACITY

From tests conducted on rolled I-section steel beams in which lateral deflection was prevented at the critical moment section, an empirical expression was developed by Kemp,²³ to fit the predictions of a model for the relationship between the available rotation capacity, R_u , and the effective lateral slenderness ratio, λ_e given by:

$$R_u = 3 \left(\frac{60}{\lambda_e} \right)^{1.5} \quad (1)$$

where:

$$\lambda_e = \frac{1}{23,948} \frac{b}{2t\epsilon} \sqrt{\frac{h}{w\epsilon}} \frac{L}{r_y \epsilon} \left(460 - 1.13 \frac{L}{r_y \epsilon} \right),$$

in which:

$$\epsilon = \sqrt{\frac{\sigma_o}{\sigma_y}},$$

and r_y is the radius of gyration of the section about its weak axis of bending. Normal stresses σ_o and σ_y represent nominal and actual yield values respectively. For typical proportions of I-shaped sections, let us assume that $w = 0.6t$ and $2b = 0.5d$. Furthermore, only one flange and one-sixth of the height of the web is assumed to be effective at the onset of the interaction of local plate buckling with lateral buckling. With these assumptions, the lateral slenderness reduces to:

$$\frac{L}{r_y} = 3.76 \frac{L}{2b} \quad (2)$$

Substituting from Equation 2 into Equation 1, λ_e can be expressed as a function of three normalized slenderness parameters given by:

$$\lambda_e = \alpha_f \sqrt{\alpha_w \alpha_t} (.07224 - .000667\alpha_t) \quad (3)$$

where:

$$\alpha_f = \frac{b}{2t\epsilon}, \alpha_w = \frac{h}{w\epsilon}, \alpha_t = \frac{L}{2b\epsilon},$$

$$\epsilon = \sqrt{\frac{300}{\sigma_y}} \text{ in S.I. units or}$$

$$\epsilon = \sqrt{\frac{43.5}{\sigma_y}} \text{ in ksi units}$$

To best fit the values of Kemp's experimental observations, the simplified empirical expression for R_u (in Equation 1) is multiplied by a correction factor, $2.277 - 0.1433\alpha_f$, that was obtained from a linear regression analysis to give a revised expression for the rotation capacity as:

$$R_u = (6.831 - .43\alpha_f) \left(\frac{60}{\lambda_e} \right)^{1.5} \quad (4)$$

Values of rotation capacity obtained with Equation 4 were then compared with only Class 1 sections or plastic design sections test data of Lukey's²⁴ and Kuhlmann's²⁵ experiments. They were found to be in good agreement; a summary of the compared values is given in Table 1. Moreover, a sensitivity analysis was performed on the selected cross-sectional dimensions. Three cases were considered; these are, namely: (1) $2b = 0.5d$ and $w = 0.6t$, (2) $2b = d$ and $w = 0.5t$, and (3) $2b = 0.5d$ and $w = 0.3t$. In all three cases, the ratio of the predicted rotation capacities to the experimentally observed ones had a mean of respectively 0.98, 1.045, and 1.04 and coefficients of variation of 0.133, 0.137, and 0.136.

DISCUSSION

From Equation 4, available rotation capacity curves (Figures 3–5) for various normalized lateral slenderness and normalized web slenderness values have been developed. Values of normalized flange slenderness were selected to have a range from five to nine while the normalized web slenderness ranged from 25 to 75; these values cover the usual spectrum of Class 1 (plastic design) sections. Three basic values for the normalized lateral slenderness, specifically eight, nine, and ten were examined. It is clearly seen in Figures 3 to 5 that the normalized lateral slenderness considerably influences the local buckling rules. Thus, for example, within the limits $\alpha_f = 8.37$ and $\alpha_w = 63.5$ specified by CSA S16.1 M-89²⁵ and $\alpha_f = 7.91$ and $\alpha_w = 78.8$ specified by AISC,²⁶ it is evident that the values of the available rotation capacity decrease by the same order of magnitude both by increasing flange slenderness and lateral slenderness. Meanwhile, R_u decreases by a lesser order of magnitude with increasing web slenderness.

If Canadian buckling limits are used as reference lines (CSA S16.1 M-89), then a careful study of Figure 4 shows that members, with a normalized lateral slenderness of nine and having plate elements satisfying the Class 1 section limits, will reach a required rotation capacity of about six. A normalized lateral slenderness of nine can therefore be taken as the critical limit which should be used in parallel with the current buckling rules in order for a section to be able to redistribute moments in accordance with plastic design. For a normalized lateral slenderness greater than nine, it is clearly shown in Figure 5 that the required value of the rotation capacity of six is reached only if more stringent buckling limits are imposed on the plate elements. Thus, for example, with $\alpha_f = 10$, a normalized flange slenderness of seven would

Reference	Spec. #	d mm	b mm	t mm	w mm	σ_y	α_f	α_w	α_l	Observed R_u	Predicted R_u
Lukey (1969)	1	250	203	10.8	7.6	277	9.06	31.4	8.2	11.8	10.49
	2	250	176	10.8	7.6	277	7.83	31.4	8.02	13.6	13.77
	3	200	74	5.28	4.4	372	7.79	50.1	7.8	10.4	10.42
	4	200	86	5.28	4.4	372	9.06	50.1	8.09	6.7	7.63
	5	250	74	5.25	4.6	372	7.79	60.7	7.26	13.7	10.61
	6	250	86	5.25	4.6	372	9.06	60.7	7.57	8.0	7.69
	7	250	90	5.25	4.6	372	9.53	60.7	7.69	6.5	6.89
Kuhlmann (1989)	8	260	141	10.2	5.5	333	7.27	44.5	11.2	5.1	4.57
	9	259	150	10	5.5	333	7.91	49.6	11.26	3.8	3.64
	10	258	160	10.4	5.5	333	8.08	49.6	11.55	3.6	3.20
	11	220	160	10	5.5	333	8.43	42.1	7.26	9.5	12.4
	12	298	160	10	6	333	8.43	52.3	6.58	12	12.99
	13	299	160	10	6	333	8.43	52.5	7.91	8.7	8.66
	14	299	160	10	6	333	8.43	52.5	9.24	7.2	5.87

be required for $\alpha_w = 55$. For the American buckling limits, it is found that a critical lateral slenderness of about eight should be used along the current rules in order to achieve a rotation capacity of about 7.5.

Interaction curves, relating flange slenderness and web slenderness for three basic values of the lateral slenderness and three required rotation capacities are shown in Figures 6 to 8. These were developed using Equation 4. As reference

lines in the interaction diagrams, vertical lines positioned at $\alpha_f = 8.37$ and 7.91 represent the CSA S16.1 M-89 and the AISC-LRFD flange buckling limits respectively, while horizontal lines positioned at $\alpha_w = 63.5$ and 78.8 denote the web buckling limits. A study of the interaction curves in Figure 8 shows that members with lateral slenderness values, α_l , less than or equal to nine, together with the Canadian code limits will develop an available rotation capacity of six. For greater

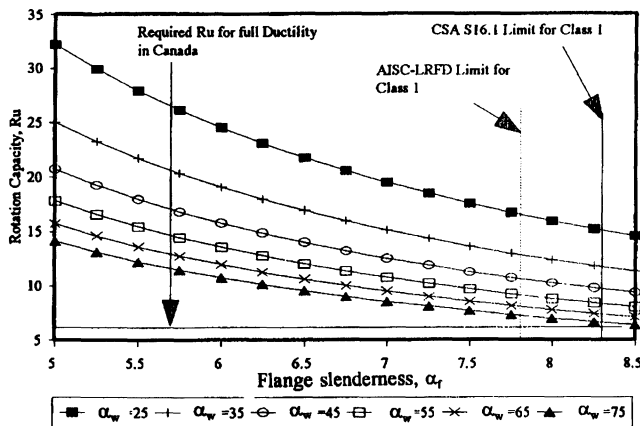


Fig. 3. Ultimate rotation capacity, R_u , for $L/B\xi = 8$.

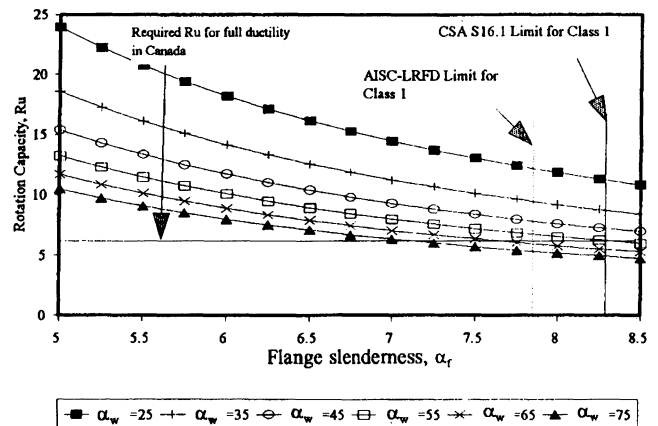


Fig. 4. Ultimate rotation capacity, R_u , for $L/B\xi = 9$.

values of the lateral slenderness, more stringent limits have to be satisfied for a member to reach an available rotation capacity of six. Interaction diagrams for $R_u = 5, 6,$ and 7 are shown in Figures 7 to 9 respectively.

EXAMPLE

For the purpose of illustration, two cantilever beam sections subjected to tip loading are selected. The length L is taken as 1,650 mm (65 in.) and $\sigma_y = 300$ MPa (43.5 ksi).

1. According to CSA S16.1 M-89 local buckling design rules where S.I. units are used, a W310x39 section has:

- $(b / 2t) = (165 / 2 \times 10) = 8.25 < (145 / \sqrt{\sigma_y}) = 8.37$
- $(h / w) = (290 / 6) = 48.33 < (1,100 / \sqrt{\sigma_y}) = 63.5$
- $(L / r_y) = (1,650 / 38.4) = 43 < (980 / \sqrt{\sigma_y}) = 56.6$

Hence, the member would satisfy plastic section and lateral support requirements in Canada.

Now, according to the procedure developed in this paper, a member having:

- $\alpha_f = b / 2t\sqrt{(300 / \sigma_y)} = 165 / 2 \times 10\sqrt{(300 / 300)} = 8.25$
(in Mpa) or $b / 2t\sqrt{(43.5 / \sigma_y)} = 8.25$ (in ksi)
- $\alpha_w = h / w\sqrt{(300 / \sigma_y)} = 290 / 6\sqrt{(300 / 300)} = 48.33$ (in Mpa) or $h / w\sqrt{(43.5 / \sigma_y)} = 48.33$ (in ksi)
- $\alpha_l = L / 2b\sqrt{(300 / \sigma_y)} = 1,650 / 38.4\sqrt{(300 / 300)} = 10$
(in Mpa) or $L / 2b\sqrt{(43.5 / \sigma_y)} = 10$ (in ksi)

would develop, according to Figure 5, an available rotation capacity of about five. This is less than the minimum required rotation capacity of six required for fully ductile structures.

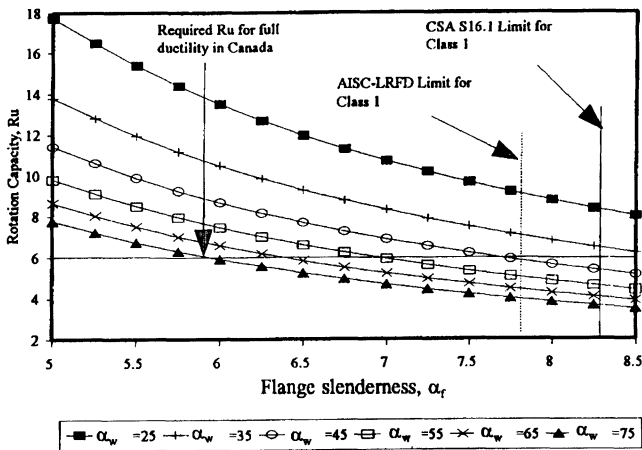


Fig. 5. Ultimate rotation capacity, R_u , for $L/B\xi = 10$.

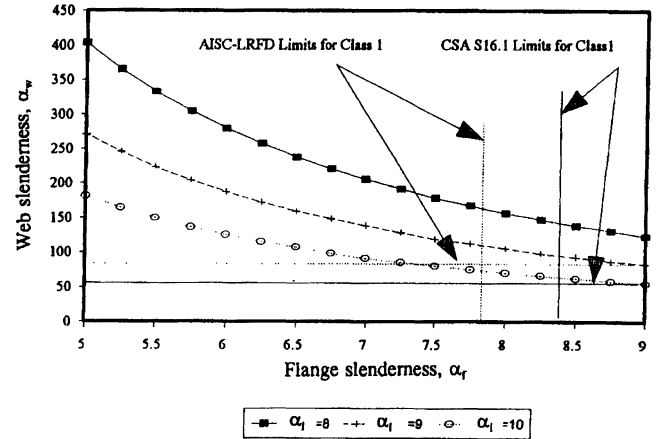


Fig. 7. Interaction diagram for $R_u = 5$.

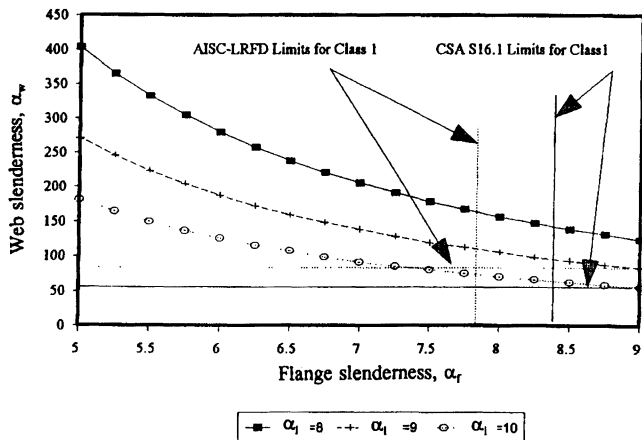


Fig. 6. Interaction diagram for $R_u = 4$.

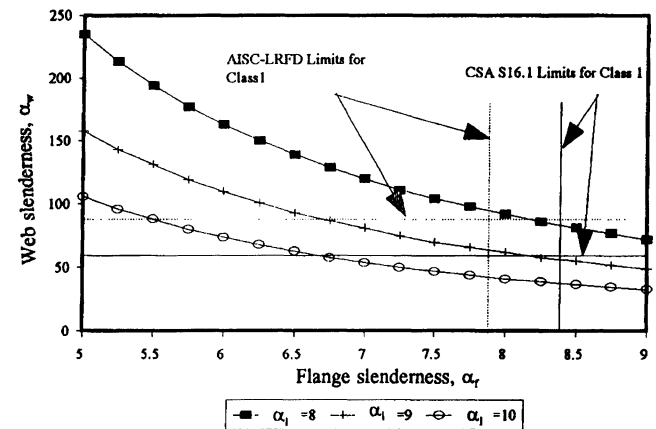


Fig. 8. Interaction diagram for $R_u = 6$.

2. Now, according to AISC local buckling design rules and S.I. units, a W310×45 section has:

- $(b/2t) = (166/11) = 7.55 < (137/\sqrt{\sigma_y}) = 7.91$
- $(h/w) = (291/7) = 42 < (1,365/\sqrt{\sigma_y}) = 78.8$
- $(L/r_y) = (1,650/38.8) = 43 < (9,473/\sigma_y) + 25 = 56.57$

Hence, the member would satisfy plastic section and lateral support requirements in the United States. Now, according to the procedure developed in this paper, a member having:

- $\alpha_f = 7.55$
- $\alpha_w = 42.0$
- $\alpha_l = 10.0$

would develop, according to Figure 5, an available rotation capacity of about six which is less than the minimum required rotation capacity of seven required for fully ductile structures. Therefore, in both cases, the member would not be able to redistribute moments in accord with plastic design unless more stringent buckling limits were applied or interaction curves such as those in Figure 5 used.

We have seen, in Figures 3 to 5, how the local buckling rules are highly influenced by the flange lateral slenderness. Further it was demonstrated through a simple example that the current local buckling limits may sometimes lead to an unconservative design. It is also seen how one can use the interaction curves to select a member with specified slenderness limits in order to achieve a chosen required rotation capacity.

CONCLUSIONS

From the study at hand, one can see that the member rotation capacity demands imposed by the seismic design codes are not always consistent with the classification of sections. It may be that member capacity values are more than ample in resisting the factored loads, however, the purpose of this study

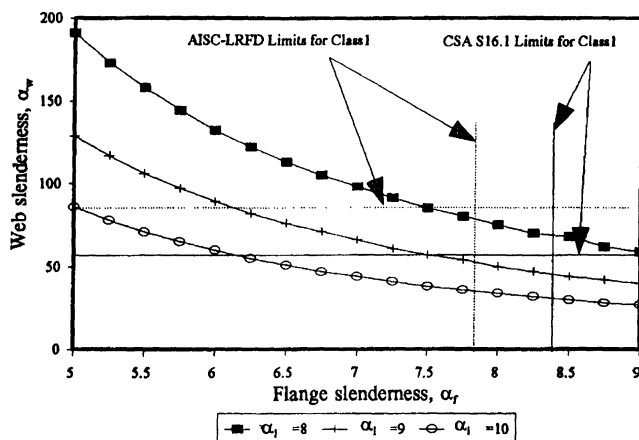


Fig. 9. Interaction diagram for $R_u = 7$.

was not to examine that issue. It was simply to accept code requirements and identify weaknesses in classification for design necessitating full ductility. Designing a structure, by employing the current local buckling rules as specified by most codes of practice, may lead to an unsafe selection of members not able to redistribute moments in accord with plastic design. An empirical expression, relating the plate slenderness with the flange lateral slenderness, is used to develop ultimate rotation capacity and interaction curves for selection of local buckling limits needed to achieve a specified required rotation capacity.

An upper limit α_l of nine is proposed as the critical limit for the flange lateral slenderness to be used with the Canadian buckling rules for beam sections required to develop a minimum rotation capacity of six. Similarly, an α_l of eight is suggested for beams required to achieve a minimum rotation capacity of seven in the United States. These values appear to permit design for full ductility. For greater values of the lateral slenderness, interaction curves are proposed to be used to reach a specified rotation capacity. The procedure can easily be applied to any standards of practice by simply replacing the buckling limits by the specified ones.

REFERENCES

1. Thurston, S. J., "Local Buckling of Universal Beam Flanges," *Bulletin NZNSE*, Vol. 15, No. 2., June 1982.
2. Beamish, M. J., "Cyclic Loading Tests on Steel Portal Frame Knee Joints," *Bulletin NZNSE*, Vol. 20, No. 1., March 1987, pp. 42–51.
3. Mitani, I. et al., "Influence of Local Buckling on Cyclic Behaviour of Steel Beam-Columns," *Proc. 6 WCEE*, Vol. 111, New Delhi, 1977, pp. 3175–3180.
4. Takanashi, et al., "Failure of Steel Beams Due to Lateral Buckling Under Repeated Loads," *IABSE*, Lisbon, 1973.
5. Kemp, A. R., "Factors Affecting the Rotation Capacity of Plastically Designed Members," *The Struct. Engineer*, Vol. 64B, No. 2., June 1986, pp. 28–35.
6. Ghobarah, A., et al., "Behaviour of Extended End-Plate Connections Under Cyclic Loading," *Engineering Struct.*, Vol. 12, Jan. 1990, pp. 15–27.
7. Walpole, W. R., "Beam Design," *Bull. NZNSEE*, Vol. 18, No. 4, 1985.
8. Lukey and Adams, "Rotation Capacity of Beams Under Moment Gradient," *ASCE*, Vol. 95, No. ST6, 1969.
9. Kato, B., "Rotation Capacity of H-Section Members as Determined by Local Buckling," *J. Const. Steel Research*, 13, 1989.
10. Kato, B., "Deformation Capacity of Steel Structures," *J. Constr. Steel Research*, 17, 1991.
11. Kuhlmann, U., "Definition of Flange Slenderness Limits on the Basis of Rotation Capacity Values," *J. Const. Steel Research*, 14, 1989.
12. AIJ LSD, *Standard for Structural Steel Structures*, Architecture Inst. of Japan, 1990.

13. Bild and Kulak, "Local Buckling Rules for Structural Steel Members," *J. Const. Steel Research*, 20, 1991.
14. Popov, E. P., "Seismic Behaviour of Structural Subassemblages," *ASCE*, Vol. 106, No. ST7, 1980.
15. Hall, W. J., *Current Trends in the Seismic Analysis and Design of Structures and Facilities*, Dept. of Civil Engng., University of Illinois, Urbana-Champaign, Ill., 1978, pp. 147–163.
16. Chopra, A. K. and Newmark, N. M., *Design of Earthquake Resistant Structures*, E. Rosenblueth, Pentech Press, 1980.
17. Fukomoto and Itoh, "Width-to-Thickness Ratios for Plate Elements in Earthquake Engineering Design of Steel Structures," *Stability and Ductility of Steel Structures Under Cyclic Loading*, by Fukomoto and Lee, CRC Press, 1992.
18. Cosenza et al., "A Rational Formulation for the q-Factor in Steel Structures," *Proc. 9th WCEE*, Tokyo, Japan, Vol. V, 1988.
19. NBCC, *National Building Code of Canada*, National Research Council of Canada, 1990.
20. NEHRP, *Recommended Provisions for the Development of Seismic Regulations for Buildings*, Building Seismic Safety Council, Washington, D.C., 1992.
21. *Uniform Building Code (UBC)*, International Conference of Building Officials, Whittier, CA, 1991
22. Uang, C. M., "Establishing R (or R_w) and C_d Factors for Building Seismic Provisions," *ASCE*, Vol. 117, No. 1, Jan. 1991.
23. Kemp, A. R. and Dekker, N. W., "Available Rotation Capacity in Steel and Composite Beams," *The Structural Engrg.*, Vol. 69, No. 5, 1991.
24. Canadian Standards Association, "Steel Structures for Buildings—Limit States Design," *CSA S16.1 M89*, Rexdale, Ontario, Canada, 1989.
25. AISC, *Seismic Provisions for Structural Steel Buildings*, American Inst. of Steel Const., Chicago, USA, 1992.

APPENDIX: NOTATION

- b = flange width
 h = web depth
 t = flange thickness
 L = length between point of inflexion and point of maximum moment
 M_p = plastic moment
 R_u = rotation capacity reached before M_p is exhausted
 R_m = material capacity
 R_s = system capacity
 r_y = radius of gyration about a section's weak axis
 w = web thickness
 α_f = normalized flange slenderness
 α_l = normalized lateral slenderness
 α_w = normalized web slenderness
 λ_e = effective lateral slenderness ratio
 θ = end rotation
 θ_p = plastic rotation
 σ_o = nominal yield stress of a prescribed value
 σ_y = actual yield stress
 μ_s = system ductility